

IAF SPACE PROPULSION SYMPOSIUM (C4)
Late breaking abstracts (LBA)

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MICROWAVE ELECTROTHERMAL THRUSTER (MET) WITH CARBON DIOXIDE (CO₂)**Abstract**

The project developed and optimised an existing (Microwave Electrothermal Thruster) MET for CO₂ to be in use with it. The MET was predominantly used with Argon and Xenon previously. The introduction of CO₂ as propellant can produce a higher Isp at the same power and is able to expel waste by using it as a form of energy. This is especially crucial for future developments of exploration upon Mars (with high CO₂ content in its atmosphere). The project is expected to: Optimise the injectors design (Number of injectors and tangential angle) and running on the propellant of CO₂, and possibly with co-reactants (E.g. Argon). Having an optimised injector and running the MET on CO₂ is expected to increase the performance and efficiency of the MET. The work is a collaboration between University of Southampton and AVS UK. The project will be due for completion in August.